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(54) **GAS BEARING DEVICE AND TURBOCHARGER**

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(71) Applicant: **mitsubishi heavy industries engine & turbocharger, LTD.**, Sagami-hara-shi, Kanagawa (JP)

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(72) Inventors: **Takaya FUTAE**, Tokyo (JP); **Shuichi ISAYAMA**, Tokyo (JP); **Toru SUZUKI**, Tokyo (JP); **Naomichi SHIBATA**, Sagami-hara-shi (JP)

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(73) Assignee: **mitsubishi heavy industries engine & turbocharger, LTD.**, Sagami-hara-shi, Kanagawa (JP)

(57) **ABSTRACT**

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A gas bearing device is a gas bearing device for rotatably supporting a rotational shaft by using a gas as a working fluid, including: a housing where the rotational shaft penetrates; an annular top foil disposed inside the housing and surrounding an outer periphery of the rotational shaft; a back spring disposed between the top foil and the housing, and having a plurality of crests contacting the top foil and a plurality of valleys contacting the housing; and a pair of snap rings fitted in shaft holes where the rotational shaft of the housing penetrates, and configured to restrict movement of the back spring in an axial direction of the rotational shaft. An inner peripheral surface of each of the pair of snap rings has self-lubricating properties.

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Feb. 26, 2021 (JP) 2021-030469

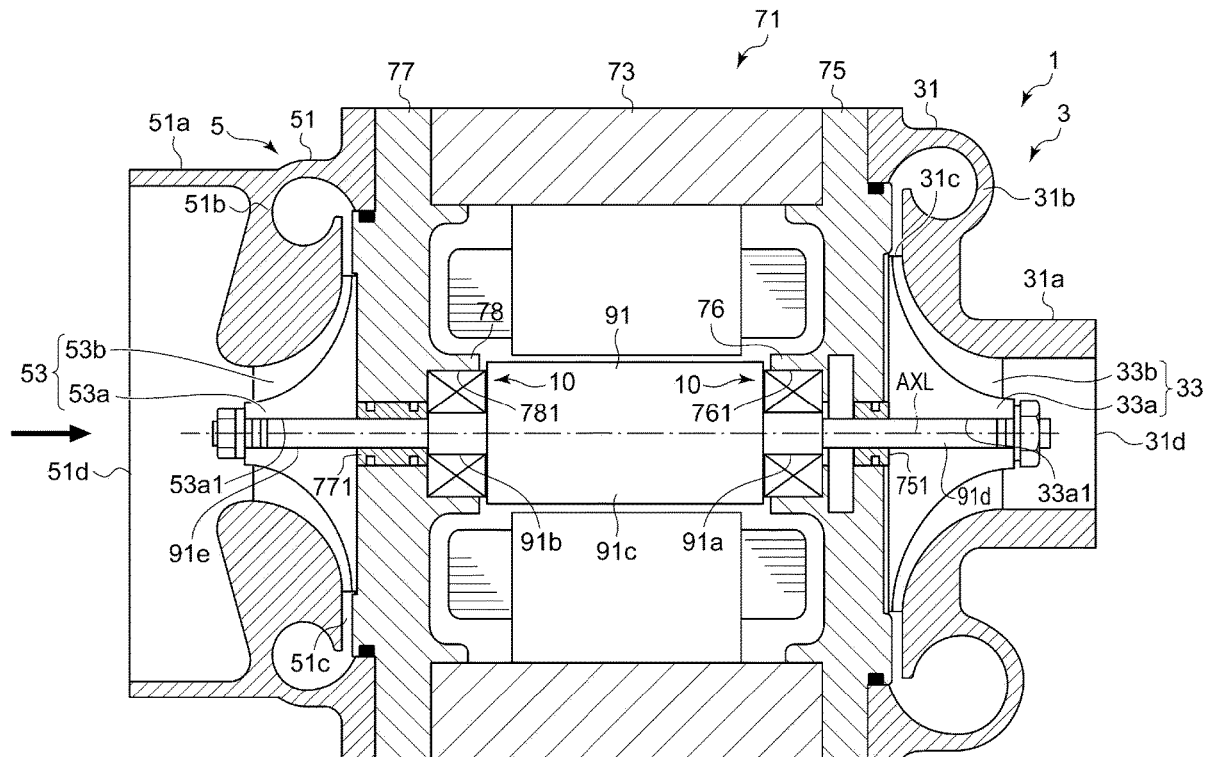


FIG. 1

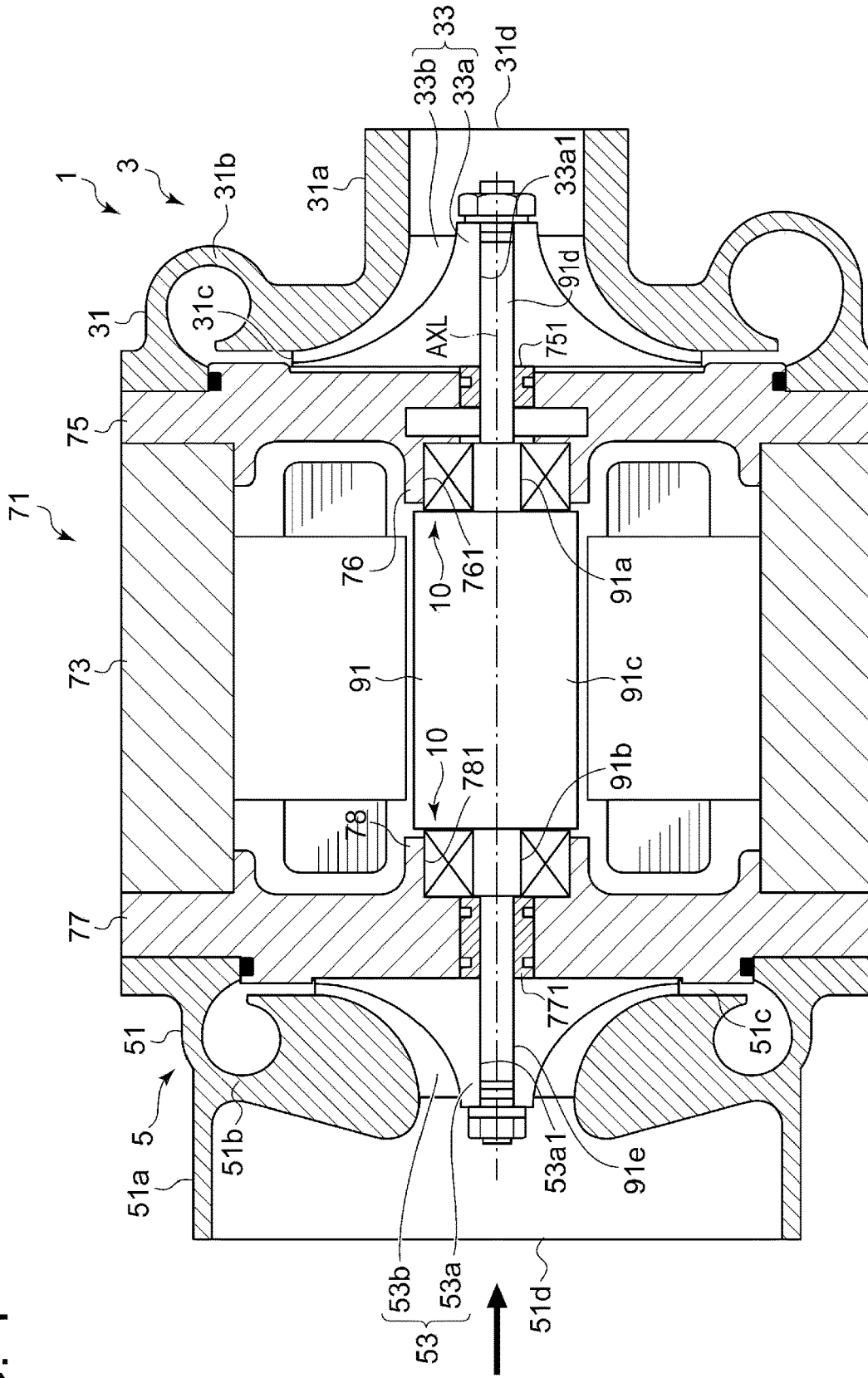


FIG. 2

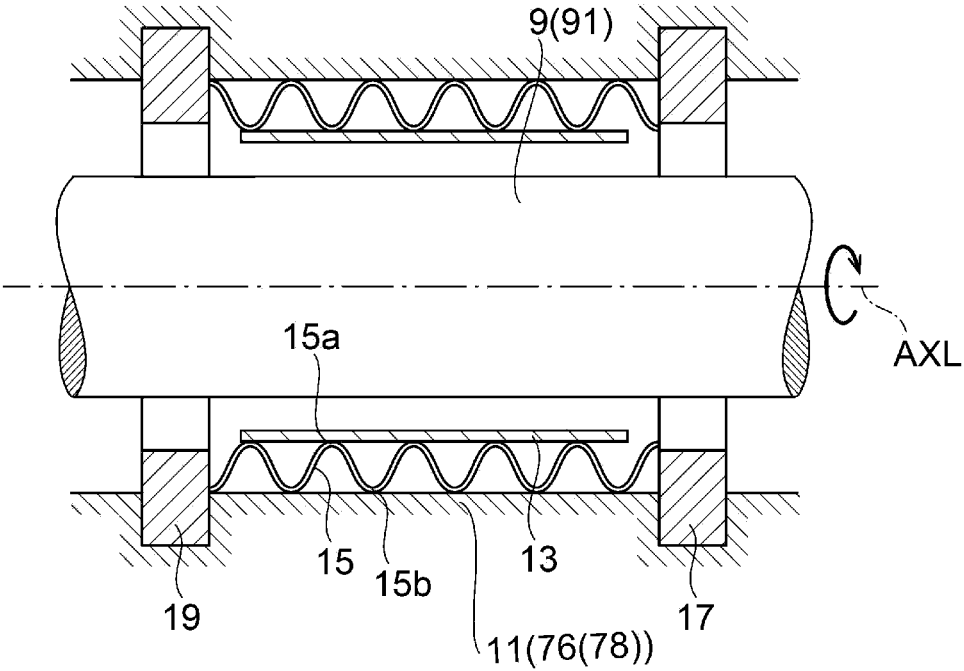


FIG. 3

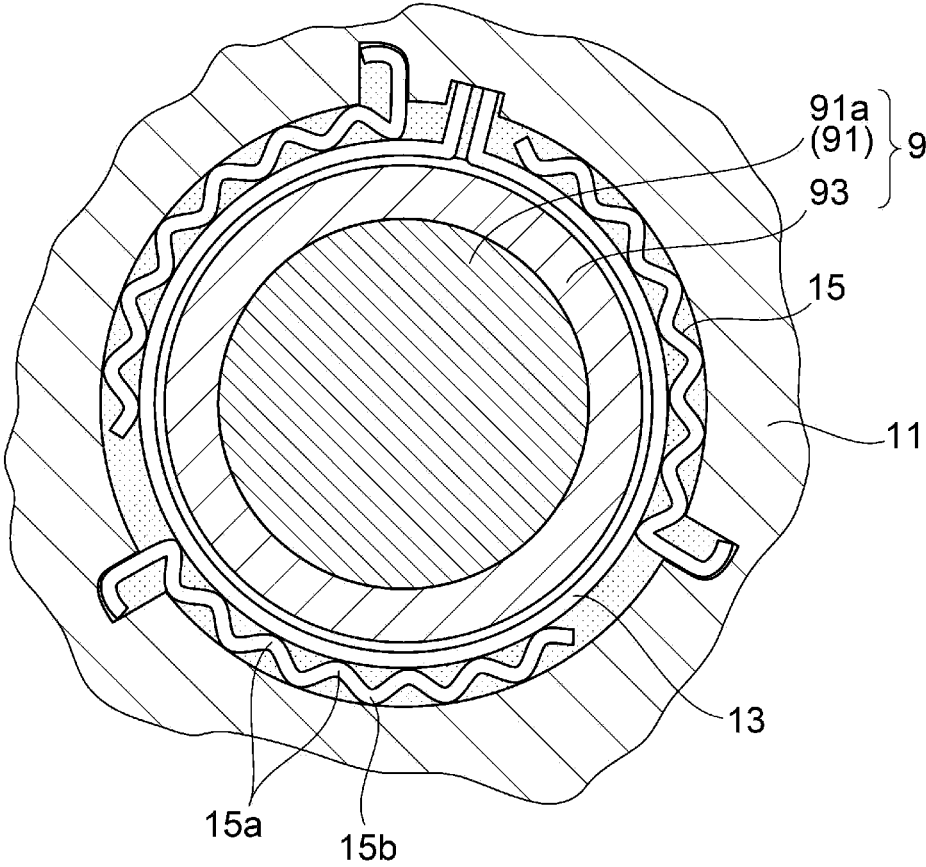


FIG. 4-1

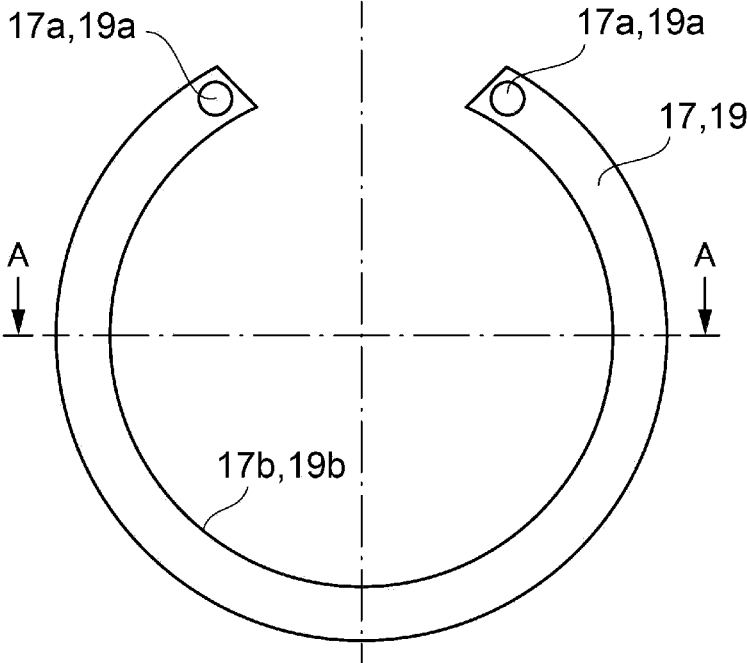


FIG. 4-2

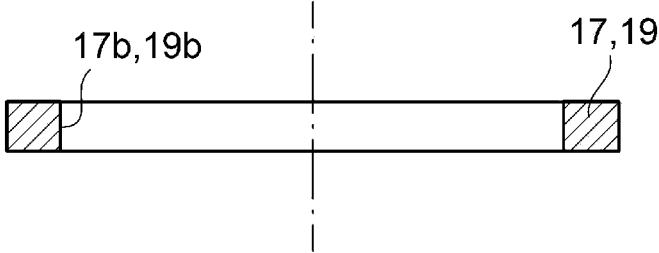


FIG. 5-1

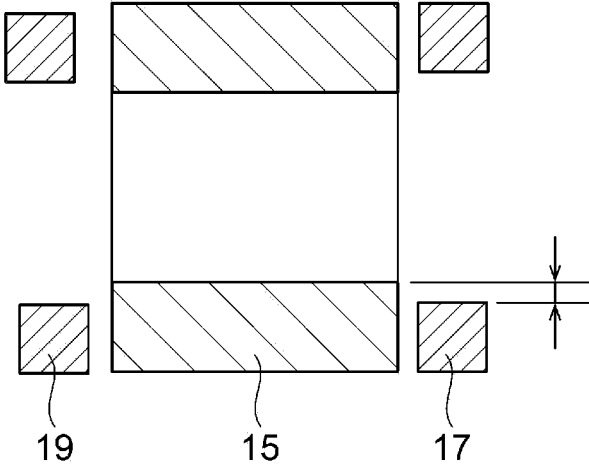


FIG. 5-2

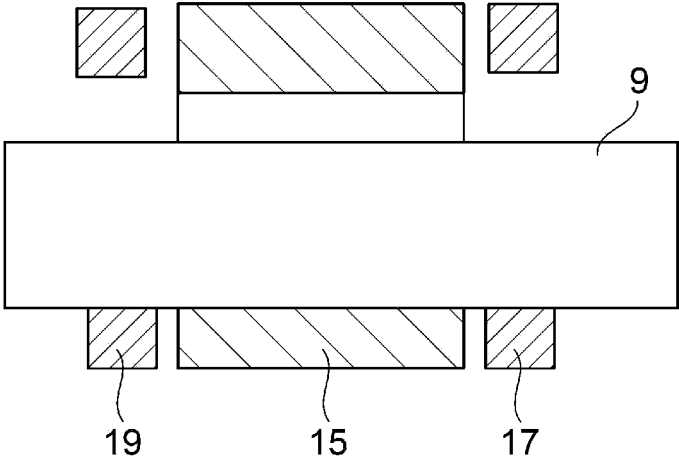


FIG. 5-3

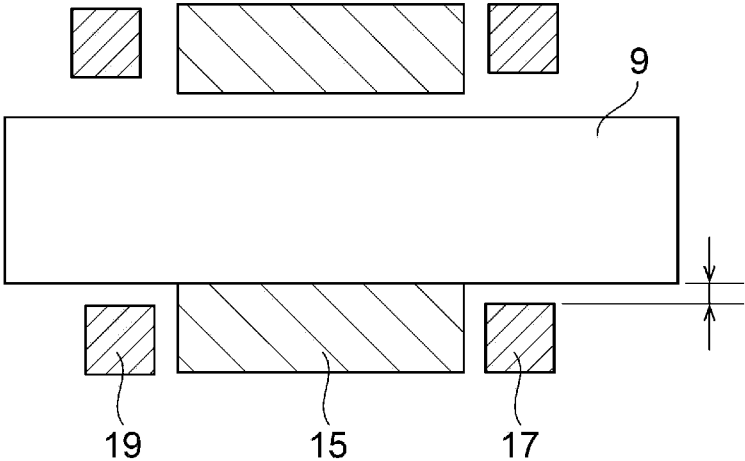


FIG. 6

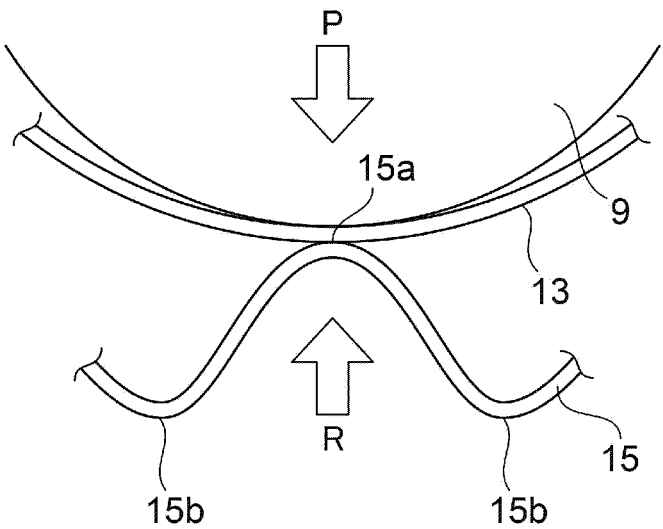


FIG. 7-1

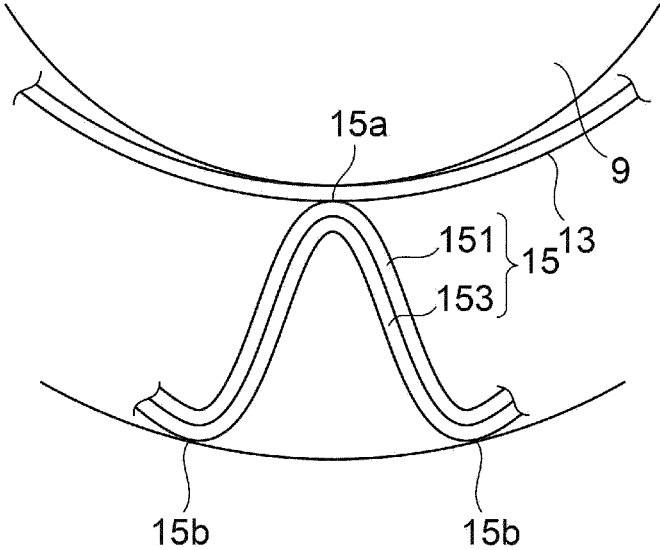
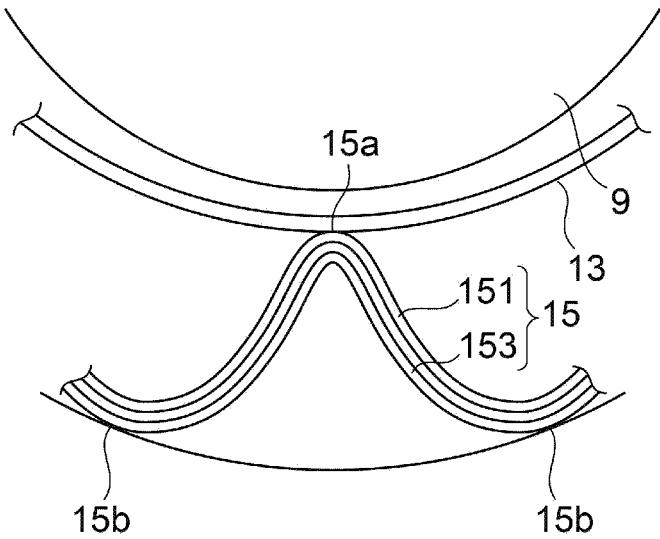


FIG. 7-2



GAS BEARING DEVICE AND TURBOCHARGER

TECHNICAL FIELD

[0001] The present disclosure relates to a gas bearing device and a turbocharger.

[0002] This application claims the priority of Japanese Patent Application No. 2020-030469 filed on Feb. 26, 2021, the content of which is incorporated herein by reference.

BACKGROUND

[0003] Patent Document 1 discloses a gas bearing device that includes a ring-shaped top foil into which a rotatable cylindrical rotational shaft is inserted and an inner peripheral surface of which is spaced from an outer peripheral surface of the rotational shaft, a damping member which is disposed on an outer periphery of the top foil and is configured to damp vibration in a direction intersecting an axis of the rotating rotational shaft, and a ring-shaped housing disposed on an outer periphery of the damping member.

CITATION LIST

Patent Literature

[0004] Patent Document 1: JP2020-122555A

SUMMARY

Technical Problem

[0005] However, in the gas bearing device disclosed in Patent Document 1, if stiffness of the damping member (back spring) is high, a gas film is not formed between the rotational shaft and the top foil during high-speed rotation of the rotational shaft, which may increase a mechanical loss. On the other hand, if the stiffness of the damping member (back spring) is low, the rotational shaft cannot rotatably be supported during low-speed rotation of the rotational shaft.

[0006] The present disclosure has been made in view of the above-described problems, and the object of the present disclosure is to provide a turbocharger and a gas bearing device capable of reliably forming the gas film between the rotational shaft and the top foil during high-speed rotation of the rotational shaft, and reliably rotatably supporting the rotational shaft during low-speed rotation of the rotational shaft.

Solution to Problem

[0007] In order to achieve the above object, a gas bearing device according to the present disclosure is a gas bearing device for rotatably supporting a rotational shaft by using a gas as a working fluid, including: a housing where the rotational shaft penetrates; an annular top foil disposed inside the housing and surrounding an outer periphery of the rotational shaft; a back spring disposed between the top foil and the housing, and having a plurality of crests contacting the top foil and a plurality of valleys contacting the housing; and a pair of snap rings fitted in shaft holes where the rotational shaft of the housing penetrates, and configured to restrict movement of the back spring in an axial direction of the rotational shaft. An inner peripheral surface of each of the pair of snap rings has self-lubricating properties.

Advantageous Effects

[0008] With the gas bearing device of the present disclosure, since the inner peripheral surface of each of the pair of snap rings has self-lubricating properties, the load of the rotational shaft need not be supported only by the back spring during low-speed rotation of the rotational shaft and the support stiffness of the back spring can be reduced. Whereby, the gas film can reliably be formed between the rotational shaft and the top foil during high-speed rotation of the rotational shaft. Further, by reducing the support stiffness of the back spring, the pair of snap rings support the rotational shaft during low-speed rotation of the rotational shaft, and since the inner peripheral surface of each of the pair of snap rings has self-lubricating properties, the pair of snap rings can reliably rotatably support the rotational shaft.

BRIEF DESCRIPTION OF DRAWINGS

[0009] FIG. 1 is a longitudinal cross-sectional view schematically showing the configuration of a turbocharger in which a gas bearing device is built according to an embodiment.

[0010] FIG. 2 is a longitudinal cross-sectional view schematically showing the configuration of the gas bearing device shown in FIG. 1.

[0011] FIG. 3 is a transverse cross-sectional view schematically showing the configuration of the gas bearing device shown in FIG. 2.

[0012] FIG. 4-1 is a front view of a snap ring shown in FIG. 2.

[0013] FIG. 4-2 is a transverse cross-sectional view (cross-sectional view taken along line A-A) of the snap ring shown in FIG. 4-1.

[0014] FIG. 5-1 is a view (cross-sectional view) conceptually showing a relationship between the snap rings and a back spring, the view showing a state before a rotational shaft is installed.

[0015] FIG. 5-2 is a view (cross-sectional view) conceptually showing the relationship between the snap rings and the back spring, the view showing a state during low-speed rotation of the rotational shaft.

[0016] FIG. 5-3 is a view (cross-sectional view) conceptually showing the relationship between the snap rings and the back spring, the view showing a state during medium-speed rotation of the rotational shaft.

[0017] FIG. 6 is a view for describing support stiffness of the back spring.

[0018] FIG. 7-1 is a view conceptually showing an example in which the back spring disposed between a top foil and a housing is composed of a bimetal, the view showing a state at low temperature.

[0019] FIG. 7-2 is a view conceptually showing an example in which the back spring disposed between the top foil and the housing is composed of the bimetal, the view showing a state at high temperature.

DETAILED DESCRIPTION

[0020] A gas bearing device and a turbocharger according to embodiments will be described below with reference to the accompanying drawings. It is intended, however, that unless particularly specified, dimensions, materials, shapes, relative positions and the like of components described in

the embodiment or shown in the drawings shall be interpreted as illustrative only and not intended to limit the scope of the present invention.

[0021] FIG. 1 is a longitudinal cross-sectional view schematically showing the configuration of a turbocharger 1 in which a gas bearing device 10 is built according to an embodiment.

[0022] The turbocharger 1 with the built-in gas bearing device 10 according to the embodiment is mounted on, for example, an automobile engine. The turbocharger 1 includes a turbine 3 which rotates with a flow of an exhaust gas, and a compressor 5 which takes in and compresses air with the rotation of the turbine 3. The turbine 3 includes a turbine housing 31 and a turbine rotor blade 33 (turbine impeller) accommodated rotatably in the turbine housing 31, and the compressor 5 includes a compressor housing 51 and an impeller 53 (compressor impeller) accommodated rotatably in the compressor housing 51.

[0023] Each of the turbine housing 31 and the compressor housing 51 is fixed by a fastening member (such as a bolt) to a bearing housing 71 in which the gas bearing device 10 is built, and the turbine rotor blade 33 and the impeller 53 are coupled to each other by a rotational shaft 91 penetrating the inside of the bearing housing 71 (gas bearing device 10). Thus, the turbine rotor blade 33, the impeller 53, and the rotational shaft 91 are disposed on the same axis AXL. The turbine rotor blade 33 is rotated by an exhaust gas discharged from the automobile engine, for example, whereby the impeller 53 of is rotated via the rotational shaft 91 and supply air to be supplied to the automobile engine is compressed.

[0024] For example, the turbine housing 31 includes a cylindrical section 31a (shroud section) for accommodating the turbine rotor blade 33, and a scroll section 31b for surrounding a part of the cylindrical section 31a on a side of the bearing housing 71. The scroll section 31b has a non-depicted inlet of the exhaust gas, and is in communication with the cylindrical section 31a via a throat section 31c. An opening of the cylindrical section 31a on an opposite side from the bearing housing 71 forms an outlet 31d of the exhaust gas.

[0025] To an opening of the turbine housing 31 on the side of the bearing housing 71, an end wall 75 (turbine-side end wall) of the bearing housing 71 is fitted. The end wall 75 of the bearing housing 71 is fastened by a fastening member (for example, a bolt) to one end portion of a peripheral wall 73 (bearing housing body) formed into a cylindrical shape, and forms a part (end wall 75) of the bearing housing 71. The end wall 75 is provided with a seal section 751, the seal section 751 is formed with a seal hole penetrating the center of the end wall 75, and the rotational shaft 91 is disposed in the seal section 751.

[0026] For example, the compressor housing 51 includes a cylindrical section 51a (shroud section) for accommodating the impeller 53, and a scroll section 51b for surrounding a part of the cylindrical section 51a on the side of the bearing housing 71. The scroll section 51b has a non-depicted outlet of supply air, and is in communication with the cylindrical section 51a via a diffuser section 51c. An opening of the cylindrical section 51a on an opposite side from the bearing housing 71 forms an inlet 51d of the supply air.

[0027] To an opening of the compressor housing 51 on the side of the bearing housing 71, an end wall 77 (compressor-side end wall) of the bearing housing 71 is fitted. The end

wall 77 of the bearing housing 71 is fastened by a fastening member (for example, a bolt) to another end portion of the peripheral wall 73 (bearing housing body), and forms a part (end wall 77) of the bearing housing 71. An annular seal section 771 is fitted in the end wall 77. The seal section 771 is formed with a seal hole penetrating the center, and the rotational shaft 91 is disposed in the seal section 771.

[0028] Inside the bearing housing 71, bearing sections 76, 78 are respectively disposed in the turbine-side end wall 75 and the compressor-side end wall 77, and the bearing sections 76, 78 are respectively formed with bearing holes 761, 781. The gas bearing devices 10 according to the present embodiment are respectively disposed as radial bearings in the bearing holes 761, 781, and the rotational shaft 91 is disposed in the bearing holes 761, 781 of the bearing sections 76, 78 while penetrating the gas bearing devices 10, respectively.

[0029] The rotational shaft 91 includes a pair of shaft sections 91a, 91b, an middle section 91c, a turbine rotor blade attachment section 91d, and a compressor impeller attachment section 91e. The pair of shaft sections 91a, 91b are sections disposed in shaft holes of the bearing sections 76, 78 while penetrating the gas bearing devices 10 respectively, and extend into the seal sections 751, 771 respectively. The middle section 91c is a section disposed between the pair of shaft sections 91a, 91b, has a larger diameter than the pair of shaft sections 91a, 91b, and is provided with a step at a boundary between the pair of shaft sections 91a, 91b. The turbine rotor blade attachment section 91d is a section disposed in an end portion on a side of the turbine 3, has a smaller diameter than the shaft section 91a, and is provided with a step between itself and the shaft section 91a. The compressor impeller attachment section 91e is a section disposed in an end portion on a side of the compressor 5, and as with the turbine rotor blade attachment section 91d, has a smaller diameter than the shaft section 91b and is provided with a step between itself and the shaft section 91b.

[0030] The turbine rotor blade 33 includes a hub 33a and a plurality of blades 33b. The hub 33a has a shape which is rotationally symmetric with respect to the axis AXL. One end side of the hub 33a is located on an outlet side of the exhaust gas, and another end side of the hub 33a is located on the side of the bearing housing 71, in a direction along the axis AXL. An outer peripheral surface of the hub 33a has a trumpet shape that widens from the one end side toward the another end side, and the hub 33a has on the another end side a back surface that faces the bearing housing 71.

[0031] The hub 33a has an attachment hole 33a1 penetrating the hub 33a along the axis AXL, and the attachment hole 33a1 has openings at the both ends of the hub 33a. The plurality of blades 33b are integrally attached to the outer peripheral surface of the hub 33a, and is arranged at a predetermined interval in the circumferential direction of the hub 33a.

[0032] The impeller 53 includes a hub 53a and a plurality of blades 53b. The hub 53a has a shape which is rotationally symmetric with respect to the axis AXL. One end side of the hub 53a is located on a side of the inlet 51d of the supply air, and another end side of the hub 53a is located on the side of the bearing housing 71, in the direction along the axis AXL. An outer peripheral surface of the hub 53a has a trumpet shape that widens from the one end side toward the another end side, and the hub 53a has on the another end side a back surface that faces the bearing housing 71 (end wall 75).

[0033] The hub 53a has an attachment hole 53a1 penetrating the hub 53a along the axis AXL, and the attachment hole 53a1 has openings at the both ends of the hub 53a. The plurality of blades 53b are integrally attached to the outer peripheral surface of the hub 53a, and is arranged at a predetermined interval in the circumferential direction of the hub 53a.

[0034] FIG. 2 is a longitudinal cross-sectional view schematically showing the configuration of the gas bearing device 10 shown in FIG. 1, and FIG. 3 is a transverse cross-sectional view schematically showing the configuration of the gas bearing device 10 shown in FIG. 2. FIG. 4-1 is a front view of a snap ring 17, 19 shown in FIG. 2, and FIG. 4-2 is a transverse cross-sectional view (cross-sectional view taken along line A-A) of the snap ring 17, 19 shown in FIG. 4-1.

[0035] As shown in FIGS. 2 and 3, the gas bearing device 10 is a bearing device for rotatably supporting a rotational shaft 9 by using a gas as a working fluid, and includes a housing 11, a top foil 13, a back spring 15, and a pair of snap rings 17, 19.

[0036] The rotational shaft 9 penetrates the housing 11. For example, the housing 11 is formed by the above-described bearing section 76 (78), and the rotational shaft 9 is formed by the above-described rotational shaft 91.

[0037] The top foil 13 is disposed inside the housing 11 and has an annular shape surrounding an outer periphery of the rotational shaft 9.

[0038] The back spring 15 is disposed between the top foil 13 and the housing 11 (shaft hole), and has a plurality of crests 15a contacting the top foil 13 and a plurality of valleys 15b contacting the housing 11.

[0039] The pair of snap rings 17, 19 are fitted in shaft holes where the rotational shaft 9 of the housing 11 penetrates, and is configured to restrict movement of the back spring 15 in the axial direction of the rotational shaft 9. The pair of snap rings 17, 19 is a snap ring for hole, is fitted in a ring groove (circumferential groove) disposed in the shaft hole where the rotational shaft 9 of the housing 11 penetrates, and is fixed by an elastic restoring force of the snap ring 17, 19. As shown in FIG. 4, the pair of snap rings 17, 19 has a partially notched annular shape, is formed such that an outer circumference and an inner circumference are concentrically disposed, and has holes 17a, 19a for tool in both end portions of the notch. An inner peripheral surface 17b, 19b of each of the pair of snap rings 17, 19 has self-lubricating properties, and is constituted by, for example, a smooth surface.

[0040] FIG. 5 is a view (cross-sectional view) conceptually showing a relationship between the snap rings 17, 19 and the back spring 15, FIG. 5-1 is a view showing a state before the rotational shaft 9 is installed, FIG. 5-2 is a view showing a state during low-speed rotation of the rotational shaft 9, FIG. 5-3 is a view showing a state during medium-speed rotation of the rotational shaft 9. Further, FIG. 6 is a view for describing support stiffness of the back spring 15. Although the top foil 13 is omitted in FIG. 5, the top foil 13 is an essential component, and the omission does not mean that the top foil 13 is not the essential component.

[0041] As described above, if the inner peripheral surface of each of the pair of snap rings 17, 19 has self-lubricating properties, as shown in FIG. 5-2, a load of the rotational shaft 9 need not be supported only by the back spring 15 and the support stiffness of the back spring 15 can be reduced. As shown in FIG. 6, the support stiffness of the back spring

15 is resistance to deformation of the back spring 15 and is a reaction force (R) obtained from the back spring 15 by the rotational shaft 9.

[0042] In the gas bearing device 10 where the support stiffness of the back spring 15 is reduced, as shown in FIG. 5-1, in the state before the rotational shaft 9 is installed, the height of the crest 15a of the back spring 15 has an equilibrium length and the height of the crest 15a of the back spring 15 is located inward of inner peripheries of the snap rings 17, 19. As shown in FIG. 5-2, in the state where the rotational shaft 9 is installed, the rotational shaft 9 is supported by the back spring 15 and the snap rings 17, 19, and the height of the crest 15a of the back spring 15 is located at the same height as the inner peripheries of the snap rings 17, 19. Even in the state during low-speed rotation of the rotational shaft 9, the height of the crest 15a of the back spring 15 is located at the same height as the inner peripheries of the snap rings 17, 19, and the rotational shaft 9 is supported by the back spring 15 and the snap rings 17, 19. As shown in FIG. 5-3, in the state during medium-speed rotation of the rotational shaft 9, the rotational shaft 9 rises from the inner peripheries of the snap rings 17, 19 and the rotational shaft 9 is supported by the back spring 15. At this time, the height of the crest 15a of the back spring 15 is located inward of the inner peripheries of the snap rings 17, 19.

[0043] With such configuration, the gas film can reliably be formed between the rotational shaft 9 and the top foil 13 during high-speed rotation of the rotational shaft 9. Further, by reducing the support stiffness of the back spring 15, the pair of snap rings 17, 19 support the rotational shaft 9 during low-speed rotation of the rotational shaft 9, and since the inner peripheral surface of each of the pair of snap rings 17, 19 has self-lubricating properties, the pair of snap rings 17, 19 can reliably rotatably support the rotational shaft 9.

[0044] For example, each of the pair of snap rings 17, 19 is composed of a self-lubricating material.

[0045] With such configuration, since each of the pair of snap rings 17, 19 is composed of the self-lubricating material, the inner peripheral surface of each of the pair of snap rings 17, 19 has self-lubricating properties. Whereby, it is possible to reduce the support stiffness of the back spring 15.

[0046] For example, each of the pair of snap rings 17, 19 is composed of a synthetic resin. For example, the synthetic resin is a high functional resin called engineering plastic, and for example, MC nylon, polyacetal (POM), etc. can be adopted.

[0047] With such configuration, each of the pair of snap rings 17, 19 is composed of the synthetic resin.

[0048] For example, the back spring 15 is composed of a spring whose support stiffness is lower at high temperature than at low temperature.

[0049] With such configuration, the support stiffness of the back spring 15 increases during low-speed rotation (during startup) of the rotational shaft 9 when the back spring 15 is at low temperature, and the support stiffness of the back spring 15 decreases at high temperature of the rotational shaft 9 when the back spring 15 is at high temperature. Thus, the rotational shaft 9 is supported by the back spring 15 during low-speed rotation (during startup) of the rotational shaft 9, and the gas film is formed between the rotational shaft 9 and the top foil 13 during high-speed rotation of the rotational shaft 9. Whereby, it is possible to reliably form the gas film during high-speed rotation of the rotational shaft 9,

and it is possible to reliably support the rotational shaft 9 during low-speed rotation during rotation.

[0050] FIG. 7 is a view conceptually showing an example in which the back spring 15 disposed between the top foil 13 and the housing 11 is composed of a bimetal, FIG. 7-1 is a view showing a state at low temperature, and FIG. 7-2 is a view showing a state at high temperature.

[0051] As shown in FIG. 7 the back spring 15 disposed between the top foil 13 and the housing 11 is composed of the bimetal formed by joining together two metal plates 151, 153 having different coefficients of thermal expansion, for example. In the example shown in FIG. 7, the metal plate 153 on a side of the housing 11 has the greater coefficient of thermal expansion than the metal plate 151 on a side of the top foil 13, and the support stiffness of the back spring 15 is higher at high temperature than at low temperature.

[0052] With such configuration, the support stiffness of the back spring 15 can be lower at high temperature than at low temperature.

[0053] The present invention is not limited to the above-described embodiments, and also includes an embodiment obtained by modifying the above-described embodiments and an embodiment obtained by combining these embodiments as appropriate.

[0054] The contents described in the above embodiments would be understood as follows, for instance.

[0055] A gas bearing device (10) according to an aspect [1] is a gas bearing device (10) for rotatably supporting a rotational shaft (9) by using a gas as a working fluid, including: a housing (11) where the rotational shaft (9) penetrates; an annular top foil (13) disposed inside the housing (11) and surrounding an outer periphery of the rotational shaft (9); a back spring (15) disposed between the top foil (13) and the housing (11), and having a plurality of crests (15a) contacting the top foil (13) and a plurality of valleys (15b) contacting the housing (11); and a pair of snap rings (17, 19) fitted in shaft holes where the rotational shaft (9) of the housing (11) penetrates, and configured to restrict movement of the back spring (15) in an axial direction of the rotational shaft (9). An inner peripheral surface of each of the pair of snap rings (17, 19) has self-lubricating properties.

[0056] With such configuration, since the inner peripheral surface of each of the pair of snap rings (17, 19) has self-lubricating properties, the load of the rotational shaft (9) need not be supported only by the back spring (15) during low-speed rotation of the rotational shaft (9) and the support stiffness of the back spring (15) can be reduced. Whereby, the gas film can reliably be formed between the rotational shaft (9) and the top foil (13) during high-speed rotation of the rotational shaft (9). Further, by reducing the support stiffness of the back spring (15), the pair of snap rings (17, 19) support the rotational shaft (9) during low-speed rotation of the rotational shaft (9), and since the inner peripheral surface of each of the pair of snap rings (17, 19) has self-lubricating properties, the pair of snap rings (17, 19) can reliably rotatably support the rotational shaft (9).

[0057] [2]A gas bearing device (10) according to another aspect is the gas bearing device (10) as defined in [1], wherein each of the pair of snap rings (17, 19) is composed of a self-lubricating material.

[0058] With such configuration, since each of the pair of snap rings (17, 19) is composed of the self-lubricating material, the inner peripheral surface of each of the pair of

snap rings (17, 19) has self-lubricating properties. Whereby, it is possible to reduce the support stiffness of the back spring (15).

[0059] [3] A gas bearing device (10) according to another aspect is the gas bearing device (10) as defined in [1] or [2], wherein the self-lubricating material is a synthetic resin.

[0060] With such configuration, each of the pair of snap rings (17, 19) is composed of the synthetic resin.

[0061] [4] A gas bearing device (10) according to another aspect is the gas bearing device (10) as defined in any one of [1] to [3], wherein the back spring (15) is composed of a spring whose support stiffness is lower at high temperature than at low temperature.

[0062] With such configuration, the support stiffness of the back spring (15) increases during low-speed rotation (during startup) of the rotational shaft (9) when the back spring (15) is at low temperature, and the support stiffness of the back spring (15) decreases during high-speed rotation of the rotational shaft (9) when the back spring (15) is at high temperature. Thus, the rotational shaft (9) is supported by the back spring (15) during low-speed rotation of the rotational shaft (9), and the gas film is formed between the rotational shaft (9) and the top foil (13) during high-speed rotation of the rotational shaft (9). Whereby, it is possible to reliably form the gas film between the rotational shaft (9) and the top foil (13) during high-speed rotation of the rotational shaft (9), and it is possible to reliably support the rotational shaft (9) during low-speed rotation of the rotational shaft (9).

[0063] [5] A gas bearing device (10) according to another aspect is the gas bearing device (10) as defined in [4], wherein the back spring (15) is composed of a bimetal formed by joining together two metal plates (151, 153) having different coefficients of thermal expansion.

[0064] With such configuration, the support stiffness of the back spring (15) can be lower at high temperature than at low temperature.

[0065] A turbocharger (1) according to an aspect [6], includes: the gas bearing device (10) as defined in any one of the above [1] to [5].

[0066] With such configuration, since the inner peripheral surface of each of the pair of snap rings (17, 19) has self-lubricating properties, it is possible to reduce the support stiffness of the back spring (15). Whereby, the gas film can reliably be formed between the rotational shaft (9) and the top foil (13) during high-speed rotation of the rotational shaft (9). Further, by reducing the support stiffness of the back spring (15), the pair of snap rings (17, 19) support the rotational shaft (9) during low-speed rotation of the rotational shaft (9), and since the inner peripheral surface of each of the pair of snap rings (17, 19) has self-lubricating properties, the pair of snap rings (17, 19) can reliably rotatably support the rotational shaft (9).

REFERENCE SIGNS LIST

| | |
|---------------|---|
| [0067] | 1 Turbocharger |
| [0068] | 3 Turbine |
| [0069] | 31 Turbine housing |
| [0070] | 31a Cylindrical section (shroud section) |
| [0071] | 31b Scroll section |
| [0072] | 31c Throat section |
| [0073] | 31d Outlet of exhaust gas |
| [0074] | 33 Turbine rotor blade (turbine impeller) |
| [0075] | 33a Hub |

- [0076] 33a1 Attachment hole
- [0077] 33b Blade
- [0078] 5 Compressor
- [0079] 51 Compressor housing
- [0080] 51a Cylindrical section (shroud section)
- [0081] 51b Scroll section
- [0082] 51c Diffuser section
- [0083] 51d Inlet of supply air
- [0084] 53 Impeller (compressor impeller)
- [0085] 53a Hub
- [0086] 53a1 Attachment hole
- [0087] 53b Blade
- [0088] 71 Bearing housing
- [0089] 73 Peripheral wall (bearing housing body)
- [0090] 75 End wall (turbine-side end wall)
- [0091] 751 Seal section
- [0092] 76 Bearing section
- [0093] 761 Bearing hole
- [0094] 77 End wall (compressor-side end wall)
- [0095] 771 Seal section
- [0096] 78 Bearing section
- [0097] 781 Bearing hole
- [0098] 9 Rotational shaft
- [0099] 91 Rotational shaft
- [0100] 91a, 91b Shaft section
- [0101] 91c Middle section
- [0102] 91d Turbine rotor blade attachment section
- [0103] 91e Compressor impeller attachment section
- [0104] Gas bearing device
- [0105] 11 Housing
- [0106] 13 Top foil
- [0107] 15 Back spring
- [0108] 15a Crest
- [0109] 15b Valley
- [0110] 151,153 Metal plate

- [0111] 17, 19 Snap ring
- [0112] 17a, 19a Hole for tool
- [0113] 17b, 19b Inner circumferential surface
- [0114] AXL Axis

1. A gas bearing device for rotatably supporting a rotational shaft by using a gas as a working fluid, comprising:
 - a housing where the rotational shaft penetrates;
 - an annular top foil disposed inside the housing and surrounding an outer periphery of the rotational shaft;
 - a back spring disposed between the top foil and the housing, and having a plurality of crests contacting the top foil and a plurality of valleys contacting the housing; and
 - a pair of snap rings fitted in shaft holes where the rotational shaft of the housing penetrates, and configured to restrict movement of the back spring in an axial direction of the rotational shaft,
 wherein an inner peripheral surface of each of the pair of snap rings has self-lubricating properties.
2. The gas bearing device according to claim 1, wherein each of the pair of snap rings is composed of a self-lubricating material.
3. The gas bearing device according to claim 2, wherein the self-lubricating material is a synthetic resin.
4. The gas bearing device according to claim 1, wherein the back spring is composed of a spring whose support stiffness is lower at high temperature than at low temperature.
5. The gas bearing device according to claim 4, wherein the back spring is composed of a bimetal formed by joining together two metal plates having different coefficients of thermal expansion.
6. A turbocharger comprising:
 - the gas bearing device according to claim 1.

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