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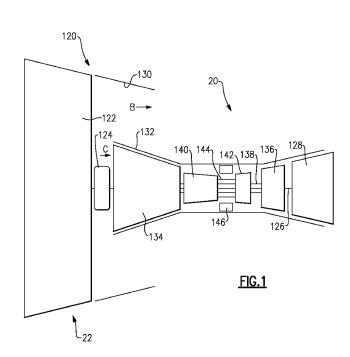
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- (88) Date of publication of the international search report:

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### (54) Title: GEARED TURBOFAN WITH THREE TURBINE SECTIONS



(57) Abstract: A gas turbine engine comprises a fan rotor configured to be driven by a fan drive turbine through a first shaft and a gear reduction. The fan rotor is configured to deliver air into a bypass duct as bypass air and to deliver core air flow into a core engine where it reaches an upstream compressor rotor. The upstream compressor rotor is configured to be driven through a second shaft by an intermediate turbine rotor. A downstream compressor rotor is configured to be driven by an upstream turbine rotor through a third shaft. An overall pressure ratio across the upstream and downstream compressor rotors is greater than or equal to about 35.0 and less than or equal to about 75.0.



### INTERNATIONAL SEARCH REPORT

### A. CLASSIFICATION OF SUBJECT MATTER

F02C 3/073(2006.01)i, F02K 3/06(2006.01)i

According to International Patent Classification (IPC) or to both national classification and IPC

#### B. FIELDS SEARCHED

Minimum documentation searched (classification system followed by classification symbols) F02C 3/073; F02C 7/36; F02C 3/06; F02C 3/10; F02K 3/06; F02C 3/04; F02G 3/00; F01D 9/04; F01D 13/00

Documentation searched other than minimum documentation to the extent that such documents are included in the fields searched Korean utility models and applications for utility models

Japanese utility models and applications for utility models

Electronic data base consulted during the international search (name of data base and, where practicable, search terms used) eKOMPASS(KIPO internal) & keywords: gas turbine engine, compressor, upstream, downstream and pressure ratio

#### C. DOCUMENTS CONSIDERED TO BE RELEVANT

Category*	Citation of document, with indication, where appropriate, of the relevant passages	Relevant to claim No.	
Y	US 2013-0318998 A1 (SCHWARZ et al.) 05 December 2013 See abstract, paragraphs [0008]-[0038] and figure 1.	1-20	
Y	US 2012-0234017 A1 (STAUBACH et al.) 20 September 2012 See abstract, paragraphs [0010]-[0013] and figure 1.	1-20	
A	EP 2592235 A2 (UNITED TECHNOLOGIES CORPORATION) 15 May 2013 See abstract, paragraphs [0020]-[0024] and figure 1.	1-20	
A	WO 2013-115871 A1 (UNITED TECHNOLOGIES CORPORATION) 08 August 2013 See abstract, paragraphs [0032]-[0038] and figure 1.	1-20	
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	Further documents are	listed in the	e continuation of Box	C



See patent family annex.

- \* Special categories of cited documents:
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- "&" document member of the same patent family

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# INTERNATIONAL SEARCH REPORT

Information on patent family members

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